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QP Code: RJ23MTECH015

GIET UNIVERSITY, GUNUPUR - 765022

M. Tech (First Semester) Examinations, January - 2024

MPCMT1010 - Composite Materials

(Manufacturing Technology)

Time	Time: 3Hrs		Maximum: 70 Marks			
(The figures in the right hand margin indicate marks.) PART – A		$(2 \times 10 = 20 \text{ Marks})$				
Q.1.	Answer all questions	C	O#	Blooms		
	William de como mano los comos circos		CO1	Level K1		
a.	What do you mean by composites?		CO1	K1		
b.	What are the commercial forms of fibers?			K2		
C.	List down the reinforcement materials and resins used in composite materials.	CO				
d.	Explain the concept of fiber orientation.	CO3		K2		
e.	Describe the relevance of the plane stress condition in the behavior of laminae.	CO3		К3		
f.	Differentiate between a cross-ply laminate and an angle-ply laminate.		CO4	K2		
g.	Write structural strain relation for laminates.		CO4	K2		
h.	Define Roving.	(CO5	K1		
i.	What is the function of caul plate?		CO5	K1		
j.	Distinguish between open mould and close mould processes.	(CO6	K2		
PART – B			(10 x 5=50 Marks)			
Answ	er ANY FIVE questions	Marks	CO#	Blooms		
2. a.	Write a brief note on engineering applications of composites.	5	CO1	Level K2		
b.	List out the reasons, why polymers are preferred in making laminated	5	CO1	K2		
	composites.					
3.a.	Write a brief note on natural and manmade composites.	5	CO1	K2		
b.	Find the stiffness matrices [A], [B] for a three ply [0/30/-45] graphite epoxy	5	CO2	K3		
	laminate. Assume each lamina has a thickness of 5mm. The properties of					
	graphite/epoxy El =181 GPa, Et =10.3 GPa, _lt = 0.28 and Glt =7.17 GPa					
4. a.	Deduce a relation between on-axis and off-axis stress transformations for a	5	CO2	К3		
	unidirectional lamina.					
b.	List the assumptions for plane stress condition.	5	CO3	K2		
5.a.	What are diluents? Explain reactive diluents for epoxy resin.	5	CO2	K2		
b.	Discuss the failure mechanisms in fibre reinforced polymer matrix composites	5				
	and describe a failure theory which is widely used for testing of polymer		CO3	K2		
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composites.

6. a.	erive the expressions to U ₁₂ , and G ₁₂ in terms of constituent properties using		CO4	К3
	Micromechanics principles.			
b.	Derive the governing differential equations for a symmetric cross ply laminated	5	CO4	K3
	plate.			
7.a.	A symmetric angle ply laminate has the following data: 4 Layers of each 0.5mm	5	CO4	K3
	thick $+45^{\circ}/-45^{\circ}/-45^{\circ}/+45^{\circ}$. El =210 GPa, Et =21 GPa, _it = 0.3 and Git =7 GPa.			
	Compute the in-plane stiffness matrix of the laminate.			
b.	Write the working principle of centrifugal casting.	5	CO5	K2
8. a.	Describe the injection molding process.	5	CO6	K2
b.	Write the difference between RTM and compression molding.	5	CO6	K2

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